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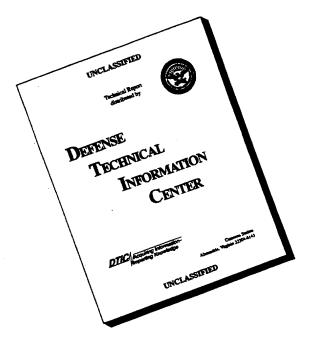
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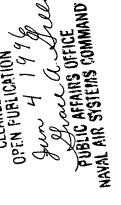
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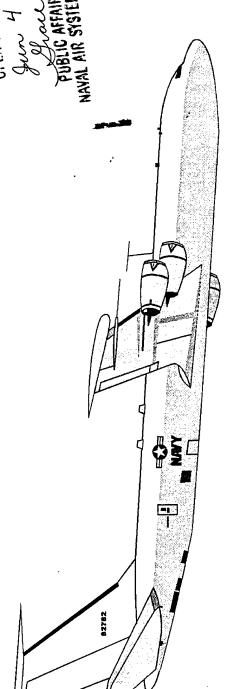
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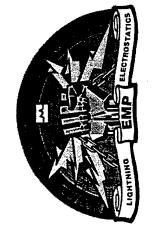
E³ HM/HS TOOLS



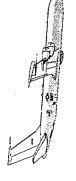




JOEL HAINES
MIKE CLELLAND



NAVAL AIR WARFARE CENTER AIRCRAFT DIVISION PATUXENT RIVER, MARYLAND 20670-5304



E-6 AIRCRAFT



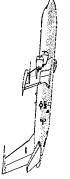
- The Design Is Based On The Commercial Boeing 707 And The Military E-3A Airframes
- Installed To Support The Take Charge And Modifications And Special Equipment Move Out (TACAMO) Strategic Communications Mission



E-6 AIRCRAFT

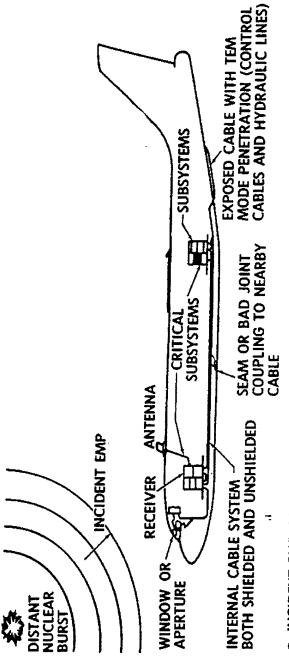


The Joint Chiefs Of Staff And The Ballistic Survivable Communications Link Between The TACAMO Mission Is To Provide A Missile Submarine Forces



EMP COUPLING

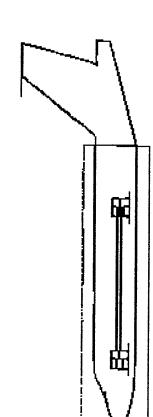




- INCIDENT EMP CAUSES CURRENT AND CHARGES ON EXTERNAL SURFACE
- SURFACE CURRENTS AND CHARGES EXCITE ANTENNAS AND INADVERTENT PENETRATIONS
 - PENETRATIONS COUPLE ENERGY TO INTERNAL CABLING
- CABLE SYSTEMS ROUTE ENERGY TO CRITICAL SUBSYSTEMS

HARDENING TECHNIOUES TYPICAL AIRCRAFT

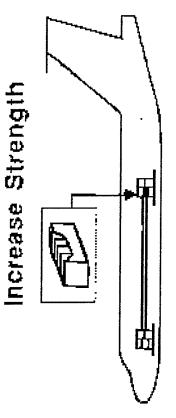
Reduce Stress



Layer 1

Hull Hardening

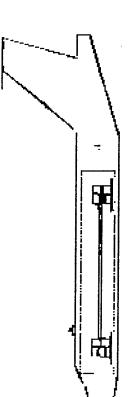
- · Antenna hardaning
- · External wire hardening
- . Mechanical penetration hardening



Layer 3

Box Hardening

- · Harden Interface circultry with alodes, filters
 - Parts centrel



Layer 2

Internal Shielding

- Wire shield rg
- . Cabinet shleiding





• BACKGROUND

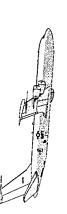
The Test Bench Nuclear Hardness (TBNH-6F) Modules Installed on the E-6A. The TBNH-6F used by the French Air Force with the mission was procured to test the Terminal Protection is a adaptation of the TBNH-160F which is equivalent to that of the E-6A.



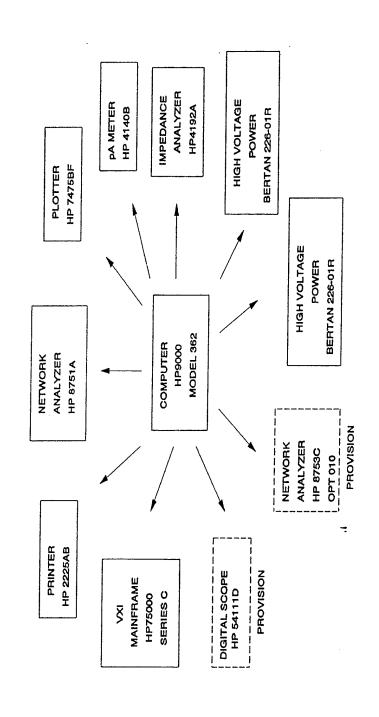


FUNCTIONAL CHARACTERISTICS

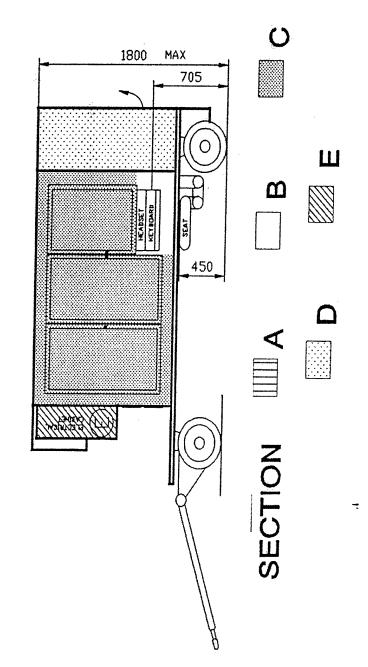
- Test Bench Allows Two Testing levels:
- Organizational Level (TPM installed on the aircraft)
- Depot Level (TPMs removed form the aircraft)
- TPM characteristics tested:
- attenuation, capacitance, inductance, resistance, firing voltage and leakage currents.

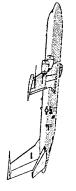














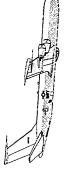
TBNH-6F CONSIST OF:

- Transportable Cart
- Air-Conditioning System
- Test Assembly



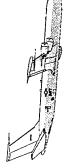


- Divided Into 5 Sections:
- Section A This Section Includes The Air-Conditioning System And Power Supply
- Retractable Seat, Computer Keyboard And The Section B - Includes A Height Adjustable, Communications With The Remote Unit) Headset Audio-Panel (For Audio
- Waterproof, Air-Conditioned Enclosure. The Cabinet Contains Three Parts Where Test Section C - This Section Is A Rigid, Fixtures Are Mounted





- Divided into 5 Sections (Cont'd)
- Electrical Connectors, And Storage Space For - Section D - Includes The Lighting System, 2 Cables
- Section E Contains A Waterproof Main Power Control Cabinet, The Interface Panel, The Main Power Interface, And 2 Drums For Main Cables





• 1st Part of Cabinet

- Picoammeter HP4140B

- Picoammeter Interface

Switching Unit HP75000-C VX 1

Interface Module Interface





- 2nd Part of Cabinet
- Bertan Power Supply
- HP 8751A Network Analyzer
- Network Analyzer Interface
- HP4192A Impedance Analyzer
- Impedance Analyzer





• 3rd Part of Cabinet

Bertan Power Supply

- HP D1194 Display

- HP A2240A Computer

Non-interruptible Power System





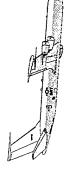
Testing System

- Controlled by a HP 9000 model 362 through HPIB bus. HP basic software is used.
- attenuation measurement (1dB accuracy) from • HP 8751A-Network Analyzer- allows TPM 2kHZ to 150 MHZ.
- HP 4140B- pA meter allows measurement of total leakage current (components, connectors, etc.), measurement of firing voltage of non-linear components up to 100 v.





- Testing System (Cont'd)
- measurement of parallel capacitors and serial HP 4192A - Impedance Analyzer- allows inductors with 1% accuracy.
- measurement of firing voltage of non-linear components for voltages higher than 100v. Bertan H.V. PWR Supply - used for



TYPICAL TEST



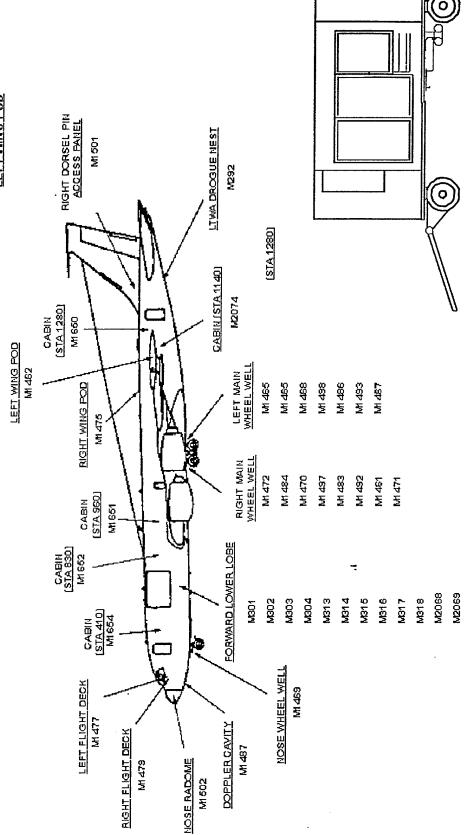
- Over 110 TPMs On Each Aircraft
- 15 30 TPMs Tested As Part Of EPM
- Only One Section Of The Aircraft Goes Through EPM At A Time
- The TBNH-6F Plots A Hard Copy And Saves All The Measured Data To A Disk



TPM LOCATIONS



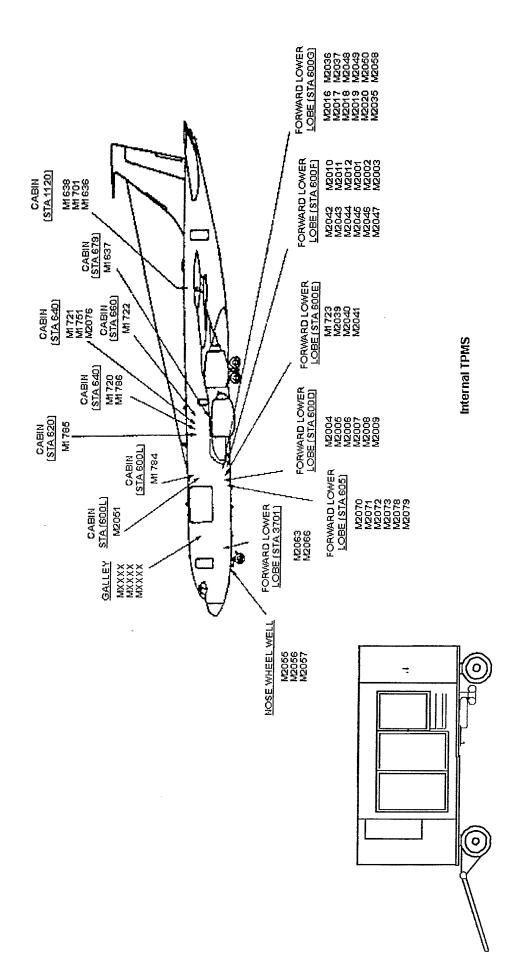
RIGHT WING POD LEFT WING POD





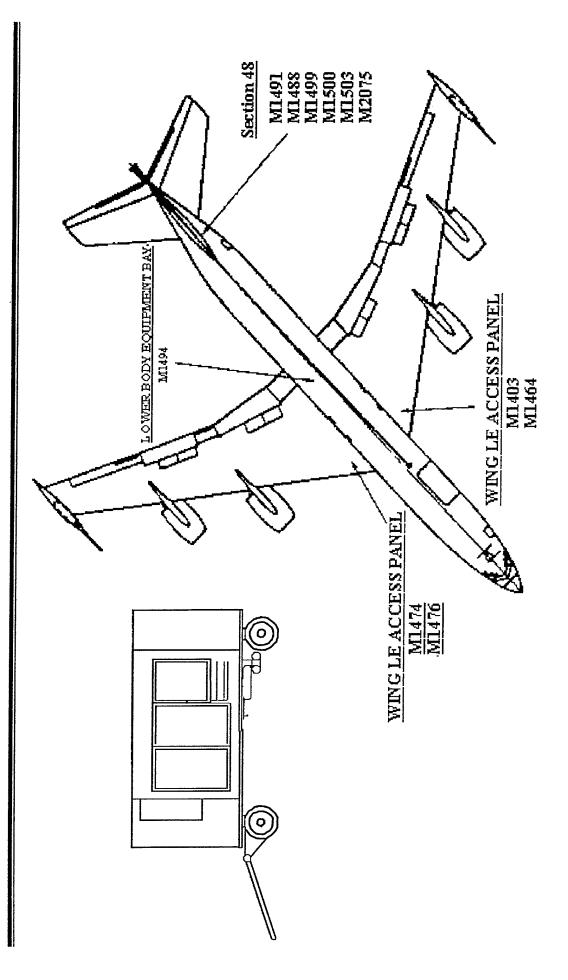
TPM LOCATIONS







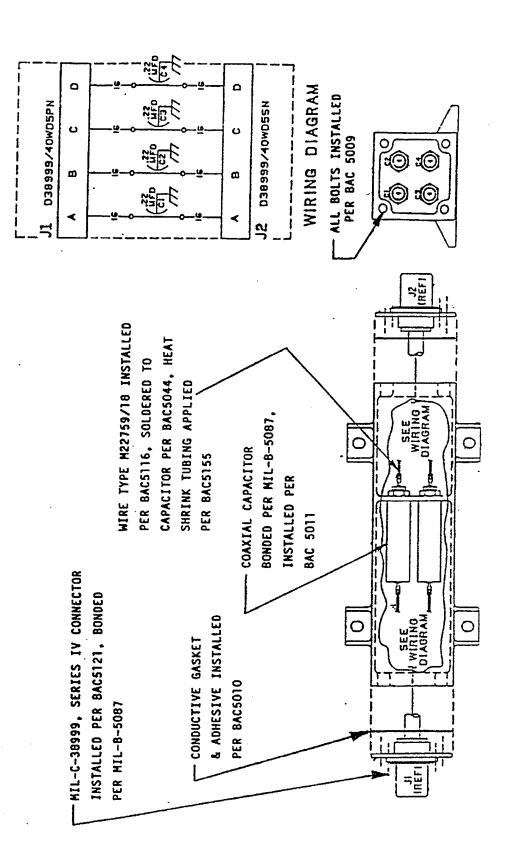








4 CAPACITOR TPM

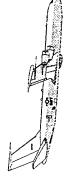






E-6 HM/HS RESULTS

- External TPM's Continue To Be A Problem
- Water Intrusion Causes Corrosion Requiring Remove and Replace
- Engineering Change In Process



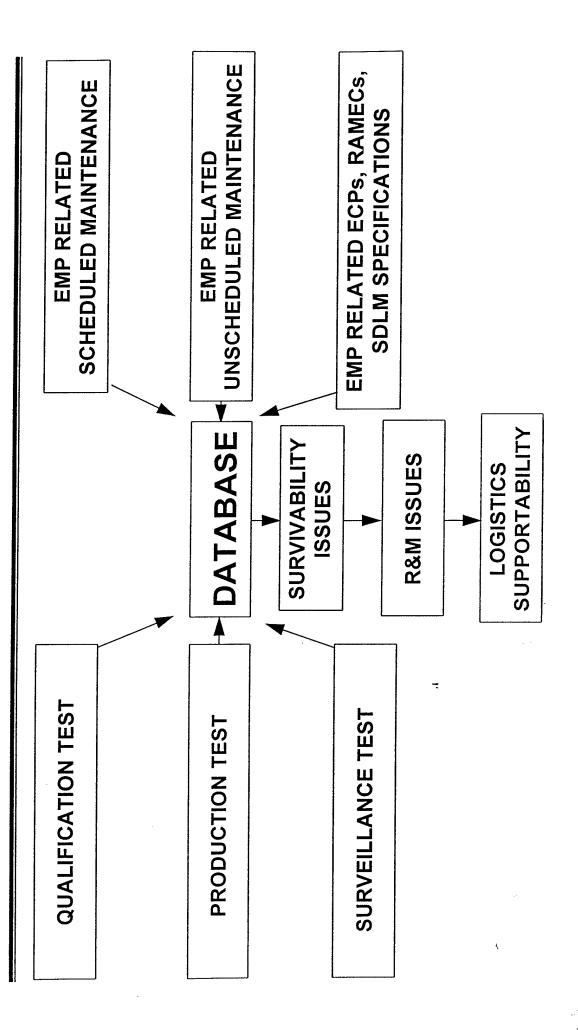
E-6 HM/HS RESULTS (Cont'd)

- 10% Of External TPM's Tested Fail "Initially"
- Excessive Amount Of Water Displacement Compound Inside Connectors
- Most TPM's Check Good After Being Cleaned, Dried & Retested
- Failure Modes Are Generally Passive Few Aircraft Downing Discrepancies



E-6 E³ HM/HS DATABASE









CONCLUSIONS

- Effective Method For Testing TPM's For The TBNH-6F Provides A Quick And The E-6 Aircraft
- The Test Set Is A Valuable Tool In The HM/HS Process